Ultra Violet Imager on Venus Climate Orbiter

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Summary

JAXA/ISAS plans to launch a mission called "Venus Climate Orbiter (VCO)" in 2008 to study the atmosphere of the Venus and to understand the origins of the atmosphere of the Earth. The scientific objectives are outlined below. To understand the mechanism of the revolution of the atmosphere To understand the structure of meridional circulation To survey the mesoscale phenomenon To understand the mechanism of cloud formation and lightning descharge To measure the ground surface radiant emittance and exploration of active volcanoes. To achieve these objectves, five cameras which are designed for differant wave length will be installed on the orbiter

and visualize the atmospheric motion on Venus with respect to the ground surface in full circumference and in three dimensions.

Ultra Violet Imager (UVI), one of VCO cameras, measures SO2 (280nm) and unknown absorber (365nm)

Optimal wavelength

Using observation of the scatterd solar ultraviolet light by topside clouds, we know distribution of unknown abserver and SO₂. The filter center is chosen as following: For unknown absorber: **365nm** is chosen because it is same wave length as PVO observation. For SO₂ distribution: Advantageous wavelength is calculated in consideration of the absorption characteristic of SO_2 . The contrast maximum is near 280 nm (It is a low wavelength side from an absorption belt peak). (Fig.4.2) 0.2 Fig.4.2: Presumption of Fig.4.1: Spectrum of (SOLOR2000) 0.2 SOLAR2000 model and the contrast by SO₂ absorption. The result 5780K black body radiation about the case of in the Venus distance. Since dl=15nm. It becomes the solar radiation spectrum has contrast maximum when the gap from that of black it is made the main



distribution on dayside cloud top by using SiCCD to estimate Cloud-tracked winds. We have designed UVI taking account of best imaging, radiation dose, weight/power limit etc., and will make preflight model since april 2005.

Science Objectives

Cloud/Haze Physics Largescale($10^3 \sim 10^5 \text{km}$) -Mesoscale($1 \sim 10^3$ km) Structures. Interaction between Lower and Middle Atmospheres

Atmosphere Dynamics

Super-rotation

Largescale/Mesoscale Wind Distribution

Planetary Waves/Gravity Waves

Distribution of

Unknowen Ultraviolet Absorber Photochemical Processes of SO₂ and H₂SO₄



Fig.1.1: Concept of the three-dimensional visualization of Venus meteorology in the proposed mission (From Japanese Venus Mission Proposal, Fig:1.3-8).

UVIObservation Outline

By using the knowledge of the past
satellite observation, in order to
attain the science purpose, it
optimizes so that efficient
observation can be performed.
Global data is obtained in short
exposure time by the two-

past o	Characteristic	Explanation
	High spatial/temporal resolution	To mesure mesoscale stractures.
	Pair Obserbation (separated from 2-4hour)	To estimate wind vector distribution
t	Long term observation	To estimate meridional circulation



Radiation Shielding

Comparing with a camera tube or a photodiode, CCDs have many merits of a low noise, low electric power, quantity sensitivity, low cost, etc. However CCDs are easy to be influenced of radiation, and it is needed to consider the measure which protects CCDs. measure which protects CCDs.

Is operation between missions (3 to 4 years) possible?

Worst case total dose: $30 \sim 50$ Krad.

- CCD characteristic is changed when 10Krad are exceeded. γ -ray: (From E2V Data sheet).
- proton: CCD has damages to the half of a pixel when 10Krad are exceeded. (The 100MeV proton irradiation experiment result in National Institute of Radiological Sciences: Fig.5.1, Fig.5.2)

 \rightarrow Imaging is possible by using dark image correcting.



Fig.5.1: A part of CCD dark image (50x50 pixels) after 100MeV proton irradiation at several situation. As the amount



Fig.5.2: Relation between total dose and dark spikes of CCD47-20 NIMO. The presumed curve is estimated by probability calculation, and fitted with experiment values. Under 10 krad(s) total dose, it is expected that the half pixel of CCD receive some damages..

dimensional solid state imager.







UV FEATURE MOTIONS..MARINER 10 OBSERVATIONS

Vehicle • •	Pioneer Venus Orbiter	Vehicle • •	Galileo
Instrument		Instrument ••	Solid-State Imag
Imaging method	Spin-scan mapping	Imaging method	two-dimensional C
• • •	by Photodiode	• • •	800x800, 8bit
Exposure Time	3.5 hour	Exposure Time	25 msec
Imaging Interval		Imaging Interval	indefinite
Spatial Resolution	∼500km, ~30km	Spatial Resolution	
Filter Center	300~390nm	Filter Center •	
S/N · ·	~ 100	S/N • •	
XOnly wind velo	city long-term	imes65 images was	taken in UV.
observation			

Vehicle • Venus Climate Orbiter Ultra Violet Imager Instrument ** two-dimensional CCD 1024x1024 12bit under 1sec Exposure Time • $2\sim4$ hours maging Interval 10km(@3.7Rv) • 280nm, 365nm (FWHM30nm) Filter Center S/N • • 100 (targeted value) •

Fig.2.1: The wind distribution that was calculated by an image out of purple of Mariner 10 (Limaye and Suomi, 1981). Such wind distribution will be estimated continually using data by UVI.

UVI Performance

observation method dayside		obserbation of the scatterd solar light by clouds top		
S	Sensor Type	The imager using the reflective refraction optical system		
Obserbation Wave length Wavelength Resolution		280nm , 365nm 30nm		
A	ngler Resolution			
	S/N	100 (Target Value)		
Ser	isor Temperature	23±3°C		
Optics	Total F Number	16		
	Synthetic Focal Length	63.3 mm		
	Aperture	Effective aperture 3.96mm		
	MTF	the Center of the Field: 0.82 the edge of the field: 0.80		
	Spot size	the Center of the Field:1 μ m the edge of the field:1 μ m		
	Glass material	Zero-dur (board)+Al+MgF2(coat)		
Detector	Туре	Si -CCD(The number of elements: 1024 × 1024)		
	Cooler	None		
	Temperature	O°C		
	Exposure Time	1sec		
	Quantum bit Number	12bit		
Filter	Filter Wheel	Position: front of lens		
	Wave Center	280nm/365nm 38.5 ϕ		
	Width	30nm		
	Shutter	Mechanical (T. B. D)		
Hood	Туре	Single Cone		
	Size	T.B.D		
	Hood Temperature	No Control		
Mass		3.2kg		
Electric	Measure	9.4W		
Power	Stand-by	4.4W		
Data rate	under AD convert	T.B.D		





increases. Each estimated total dose: (a): 12.8, (b): 72.8, (c): 322.8, (d): 2332.8 rad.

How should CCD be protected?

- Considering the structure of $\overline{C}CD47$ -20, the proton of several MeV energy gives maximum damage to the CCD (Fig.5.3).

- The CCD should be protected from the proton below this energy. - The solar proton flux at the quiet time has an extremum near GeV energy (Fig.5.4).
- It is conversely disadvantageous if the proton near an extremum becomes near a number MeV grade as a result of protection.
- The characteristic of the quality of a shield materials (Dale et al., 1993) Al:shield is more effective per unit mass.
 - Ta:shield is more effective per unit thickness
- Thickness equivalent to aluminum 10mm is an ideal shield.
- When weight-restrictions are also taken into consideration, Minimum thickness needed is equivalent to 3mm aluminum(Fig.5.5).



Fig.5.4: The proton flux around Venus which is estimated using the CREME model. At the time of a flare event, it is monotonous reduction as it becomes high energy. At the time of quiet time, it has an extremum near GeV energy.

Future schedule



Fig.5.3: The Monte Carlo simulation result by TRIM. CCD47-20 is calculated as a target which consists of substances of six layers. The proton energy which gives the maximum damage is about 1 MeV. Left: Ranges at the simulation of irradiating 10000 1MeV protons perpendicularly from the front of CCD, Right: The situation of the damage caused to CCD.





Fig.5.5: Ranges of electron and proton in aluminum. With about 3mm aluminum, CCD can be shielded from the proton of 25MeV.

Space EnvironmentInformation System



Fig.3.1: Above) Ultra Violet Imager. Below) UVI Optical system.



Component	Module	Element	Observation(W)		Standby (W)	
UVI	Hood		.0.0		0.0	
	Sensor	Detector		3.00		1.10
		Filter Wheel	3.0		1.1	
		Preamp				
	Sensor Controller	ADC Board	2.0	6.4	1.0	3.3
		TMG Board	1.5		0.8	
		Power Supply Board	2.9		1.5	
合計				9.37		4.35
UVI Electric Power Detail						

ood Brack

Filter Whee tor+Bracl

Circuit PWB+E (Circuit shackle



- It will be a start schedule about creation of an engineering model from April, 2005. Development of a driver is needed.
- OFrom the thermal demand of a satellite system, it became clear recently that a radiator may be needed for UVI. In this case, certain kind of lightweightizeing is required.

 \supset Examination of the calculation algorithm of a wind vector distribution, and program creation will be needed.

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